

Jason-3 Status



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Mission Background



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Mission Summary

Science Measurements

Global sea surface height to an accuracy of \leq 4 cm every 10 days, for determining ocean circulation, climate change and sea level rise

Mission Objectives

- Provide continuity of high precision ocean topography measurements beyond TOPEX/Poseidon , JASON-1 and JASON-2
- Provide a bridge to an operational mission to enable the continuation of multi-decadal ocean topography measurements

Instruments

• Core Mission:

- Poseidon-3B Altimeter
- DORIS (Precise Orbit Determination System)
- Advanced Microwave Radiometer (AMR)
- GPS Payload (GPSP)
- Laser Retro-reflector Array (LRA)
- Passengers:
 - JRE (Carmen3 + LPT)



Mission Overview

- Launch Date: March 2015
- Launch Vehicle: Falcon 9 (SpaceX)
- Proteus Spacecraft Bus provided by CNES
- Mission life of 3 years (goal of 5 years)
- 1336 km Orbit, 66º Inclination

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Jason-3 System Elements





Changes and new features wrt OSTM/Jason-2 (1)

- System : AMR in-flight cold-space calibration
 - Lisbon OSTST recommendation, San Diego OSTST decision
 - Satellite pitch maneuvers (80° off nadir).
- This change is completed and has been tested in 2014.





Changes and new features wrt OSTM/Jason-2 (2)

- Satellite
 - Slight modification of satellite OBSW (Tx OFF for safety improvement, EDAC robustness), PIM structure panels.

Completed and validated



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Changes and new features wrt OSTM/Jason-2 (3)

- POS3B (Altimeter)
 - Implementation of a single mode with on-board automatic transitions between DIODE/DEM tracking and autonomous tracking, with respect to the satellite position.
 - POS3B DEM upload is now possible without mission interruption.

Completed

- See J.D. Desjonqueres
- Poster : JA2 In-Flight Diode/DEM tracking mode results



>OSTST meeting

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Changes and new features wrt OSTM/Jason-2 (4)

- DORIS
 - New generation DGXX-S taking into account lessons learned from Jason-2
 - Change of DORIS antenna location for compliance with each potential launch vehicle while waiting for the selection
 - Improvement in modeling the Solar Panels position
 - New data in TM allowing "pole product" generation

Completed





Changes and new features wrt OSTM/Jason-2 (5)

- AMR (Radiometer)
 - Mostly recurring design with improvement of the instrument thermal control and stability (lesson learned from Jason-2 experience)

Completed





Changes and new features wrt OSTM/Jason-2 (6)

• GPSP

- Different receiver but with same basic design as on JASON-1/2
- Not mission critical but applying further updates for radiations hardened parts and shielding

Completed





Changes and new features wrt OSTM/Jason-2 (7)

- Launch vehicle : Falcon 9 (SpaceX)
- New Payload Processing Facility (PPF) at Vandenberg : SpaceX PPF
- Environment assessment based on launcher flight data and on Mech Tests results.
 Launcher compatibility demonstrated in summer 2014 : completed
- Launch Campaign preparation in progress for a launch planned end of March 2015





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Changes and new features wrt OSTM/Jason-2 (8)

- Ground System
 - Capability to operate simultaneously JA2 and JA3
 - Addition of stations for the "tandem flight" phase : Barrow (NOAA) and Usingen2 (EUM)
 - JASON-2 and JASON-3 operations "merging" considered after the launch

Completed

- Product Processing :
 - Development of a "numerical retracking" to be used for Jason-3 OFL products (first in PEACHI) allowing to take into account the actual instrument features before launch and in-orbit and to better estimate the low sea states.



See F. Boy's Presentation "Towards the Jason-3 waveform processing: assessment of the numerical retracking performances" (Boy et al.)

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Jason-3 Project Status : Significant events since Boulder OSTST

- March Dec 2013 : 4 partner Ground system tests for Technical Qualification completed : successful
- Mid Jan 2014 : NOAA FY2014 budget reduction
- End Feb 2014 : Launch date postponed by 1 month (contractual option exercised by NOAA). Launch date : March 31, 2015
- Feb-Mar 2014 : Sat Environmental Tests (Mech, Launcher I/F) : successful
- Jun-July 2014 : Sat Thermal Vacuum Tests : successful
- Aug 2014: Sat EMC Tests in CATR : successful
- Sept mid Oct 2014 : Sat Final Functional Tests : successful



Jason-3 Satellite AIT - 1

• Satellite in Environmental Tests





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• Satellite in Thermal Vac Chamber





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• Satellite in EMC Tests



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Jason-3 Satellite AIT - 4

• Satellite AIT team



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Jason-3 Ground System Status

- Ground Systems
 - CNES, EUM, NOAA, JPL : OK
 - All the 4 partner ground systems are implemented and validated
 - Preparation of operations : OK
 - Products : same as Jason-2
- 4 Partner tests
 - 4 partner "Compatibility Tests" and "Technical Qualification tests" are successfully completed
 - Operational Qualification tests run :
 - LEOP tests : successful
 - "Long Term Routine" tests : successful
 - First Dress rehearsal already exercised in Oct 2014 : successful



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Jason-3 Project Status : Remaining activities

- Satellite activities :
 - Satellite SQR Review : 18-20 Nov 2014
 - "Satellite Final Preparation" before Satellite shipment to VAFB in Feb 2015
 - Launch Campaign : mid Feb end March 2015
- Ground System final preparation :
 - Dress Rehearsals : between Jan 2015 and March 2015
 - 2GHz Network Rehearsals (RGR) : Feb and March 2015
- System activities :
 - JA3 Performances Assessment and CALVAL Plan KP : Dec 2, 2014
 - Mission ORR Review : end Jan 2015
- Launch and LEOP phase : end of March 2015

ason-3 Orbit Acquisition Strategy

- Tandem flight with Jason-2 planned for both altimeters cross-calibration purposes
- Jason-3 final orbit characteristics :
 - same ground tracks as Jason-2
 - between 1-10 minutes ahead/behind Jason-2
- Injection orbit :
 - 25 km below the nominal Jason-3 orbit to avoid polluting the operational orbit and to avoid to cross the Jason-2 orbit (and Ja-1)
 - duration for station acquisition and number of maneuvers depends on the launch date (day number in the Jason cycle) and on the launcher dispersions

Target duration for station acquisition : max





Jason-3 Phases

Phase	Spacecraft activities	Leader
LEOP	3-5 days: LEOP S/C & instruments functional (nom/nom mode)	CNES w/ support of EUMETSAT, NASA, NOAA
ASSESSMENT	4 weeks max: Orbit acquis Orbit acquisition key point S/C on final orbit – Jason2 & 3 tandem flight	CNES w/ support of EUMETSAT, NASA, NOAA
	2 months Assessment In Flight Assessment meeting Fully assessed S/C on final orbit S/C & GS nominal operations	
OPERATIONAL	2 months max HandOver review S/C Operations Handed Over to NOAA Till S/C decommissioning	CNES w/ support of EUMETSAT, NASA, NOAA NOAA w/ support of EUMETSAT, NASA, CNES
Phase	Products activities	Leader
VERIFICATION	Start of cycle 1 5 months max: NRT Verif First verif workshop	CNES & NASA w/ support of NOAA, EUM and Pls
	Jason2 & 3 tandem flight 10 months max: OFL Verif Jason-2 / 3 Inter Calibration Key Point Final Verif OFL products reprocessin	g
OPERATIONAL	NRT products generation & dissemination Till S/C decommissioning OFL products generation Till S/C decommissioning OFL products disseminatio Till S/C decommissioning	EUMETSAT & NOAA CNES CNES CNES & NOAA
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Conclusion

- Jason-3 development is nominal at satellite, instruments and ground levels
- Fruitful cooperation between all the project teams (CNES, EUMETSAT, NASA, NOAA)
- Final stretch of activities before JA3 Launch !!!!!

