

Precise Orbit Determination of DORIS satellites by CNES/CLS IDS Analysis Center in the frame of the next ITRF

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### Outline

Models and Standards Update

- □ Processing Strategy update
- POD resultsOPR and DORIS RMS of fit

□ Status of POD for Jason-2/3 and Sentinel-3A/3B satellites

- DORIS RMS of fit
- Independent SLR RMS of fit
- Comparison to external orbits

□ Impact on the multi-satellite solution

□ Summary



## **Models/Standards update**

#### □ Models and standards recommended by IERS

Models/Standards		OLD	NEW
Earth rotation	Mean pole	IERS2010	Linear mean pole from updated IERS conventions
	Subdaily pole model	Previous IERS convention	Desai & Sibois from updated IERS conventions
Gravity model	Time variable gravity field	EIGEN-GRGS.RL03	EIGEN-GRGS.RL04
	Oceanic/Atmospheric gravity Dealiasing Products	Νο	AOD 1B RL06 (GFZ)
Ocean tides	Station displacements (ocean loading)	FES2012	To be implemented FES2014b
	Gravitational attraction	FES2012	FES2014b



## **Processing strategy update**

#### □ Processing strategy (GINS/DYNAMO software)

Theme	OLD	NEW
Attitude modelling (Spacecraft + Solar array)	Attitude model for all satellites	Quaternions for Jason-2 and Jason-3 Quaternions for s3a, s3b and cs2 (planned)
Coefficient Solar Radiation pressure Cr	Satellite dependent estimated and fixed	Satellite and time dependent Adjusted per arc planned
Estimated measurement parameters	One frequency bias per pass	One frequency bias <b>and drift for SAA</b> stations per pass (for Jason-2 and Jason-3)
Elevation cut-off and data downweighting	Cut-off 12° downweighting: elev²/400 for elev < 20°	Cut-off 10° CNES downweighting law (planned)
Integration Step Size	60 sec	30 sec
Estimation of the DORIS Center Of Phase locations	NO	Sentinel-3A/3B, Envisat Others?



## Status of POD for the currently flying DORIS satellites

#### **POD Summary**

DORIS RMS of fit and OPR Acceleration Amplitude / Radiation pressure coefficient

DORIS SATELLITE RMS		OPR amplitude average (10 <sup>-9</sup> m/s <sup>2</sup> )		Solar radiation
(mm/s)	Along-track	Cross-track	coefficient Cr	
Cryosat-2	0.348	2.8	2.6	1.00
HY-2A	0.346	0.46	3.3	0.86
SARAL	0.340	1.3	2.7	1.00
Jason-2	0.338	3.8	2.6	0.97
Jason-3	0.360	0.9	2.3	0.99
Sentinel-3A	0.371	2.3	1.6	1.00
Sentinel-3B	0.389	1.4	1.8	1.00

Average (128 weeks) (from January 2017 to June 2019) 54 weeks for Sentinel-3B

• For the two directions, Along-track and Crosstrack, the mean amplitudes are lower than 4x10<sup>-9</sup> m/s<sup>2</sup>, reflecting a satisfying level in the modeling of the satellite macromodels and the attitude law

But could be improved by adjusting a Cr/arc in particular for Jasons satellites



#### DORIS RMS of fit



- For Jason-3, the level of DORIS RMS residuals is slightly higher compared to Jason-2, explained by its higher sensitivity to the South Atlantic Anomaly (SAA).
- There is a ~59 days periodic signal for both satellites (draconitic period).



#### Estimation of the beacon frequency Bias+Drift on SAA station per pass

Impact on the precise orbit

Classical processing: one Frequency Bias adjusted per pass for all stations With strategy: Frequency Bias+Drift adjusted per pass.

#### Average (128 weeks) (from January 2017 to June 2019)

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DORIS RMS of fit differences per station



The level of DORIS RMS residuals is slightly higher for Sentinel-3B.



# □ Comparison to CNES (POE-E/F), GSFC (1808a+DPOD2014) and JPL (jplgpspoe) orbits Independent SLR RMS of fit



 The SLR RMS residuals on Jason-2 and Jason-3 orbits are higher (few mm) to the other orbits evaluated (GPS+DORIS, DORIS+SLR and GPS-only orbits)



# Comparison to CNES (POE-F) orbits Independent SLR RMS of fit



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 The SLR RMS residuals on Sentinel-3A and Sentinel-3B orbits are also higher compared to the CNES-POE-F orbit (GPS+DORIS)

# □ Comparison to CNES (GDR-E) and GSFC orbits Jason-2 orbit differences

(from January 2017 to June 2019)

RMS and Avg. Radial orbit differences (in cm) 2 1.75 1.5 RMS (cm) 1.25 0.75 0.5 0.25 0 09-01-17 09-05-17 04-01-18 04-05-18 01-09-18 30-12-18 29-04-19 27-08-19 06-09-17 0.10.05 CNES-GDR-E Average (cm) 0 GSFC -0.05 -0.1 -0.15 -0.2 -0.25 -0.3 09-01-17 09-05-17 06-09-17 04-01-18 04-05-18 01-09-18 30-12-18 29-04-19 27-08-19

**Reference orbit = GRG orbit** 

#### Jason-2 orbit differences (cm)

Difference	Radial RMS   AVG	Cross-track RMS   AVG	along-track RMS   AVG
CNES-GDR-E	0.80   -0.13	1.9   0.12	2.3   0.21
GSFC	0.76   -0.08	2.4   0.24	2.6   0.34

There is a good agreement between GRG orbit and external orbits CNES-GDR-E and GSC (~0.8 cm RMS)

There is a 59 days periodic signal in the radial component and small bias



## □ Comparison to CNES (POE-F), GSFC and JPL orbits Jason-3 orbit differences

(from January 2017 to June 2019)



**Reference orbit = GRG orbit** 

Jason-3 orbit differences (cm)			
Difference	Radial RMS   AVG	Cross-track RMS   AVG	along-track RMS   AVG
CNES-POE-F	0.70   -0.08	2.5   0.18	2.3   0.60
GSFC	0.74   -0.08	2.8   0.21	2.6   -0.30
JPL	0.62   0.05	2.4   0.17	2.2   0.70

There is a good agreement between GRG orbit and external orbits (~0.7 cm RMS)

- There is a 59 days periodic signal (low amplitude) in the radial component and a small bias (< 1mm) (draconitic period)
- There is also an along-track bias higher with GPS orbits



#### Comparison to CNES (POE-F) orbits Sentinel-3A orbit differences

(from January 2017 to June 2019)



Reference orbit = GRG orbit

Sentinel-3A orbit differences (cm)

Difference	Radial	Cross-track	along-track
	RMS   AVG	RMS   AVG	RMS   AVG
CNES-POE-F	0.79   -0.04	1.7   0.11	2.1   -0.29

There is a good agreement between GRG orbit and CNES-GDR-E (~0.7 cm RMS)



#### Comparison to CNES (POE-F) orbits Sentinel-3B orbit differences

(from July 2018 to June 2019)

#### RMS and Avg. Radial orbit differences (in cm)



Reference orbit = GRG orbit

Sentinel-3B orbit differences (cm)

Difference	Radial	Cross-trk	along-trk
	RMS   AVG	RMS   AVG	RMS   AVG
CNES-POE-F	0.92   -0.04	1.9   -0.13	2.5   0.10

• There is a good agreement between GRG orbit and CNES-GDR-E (< 1 cm RMS)



#### □ Comparison to CNES (GDR-E) and GSFC orbits Mapping radial orbit differences

Average (128 weeks) (from January 2017 to June 2019)

**REF = GRG orbit** 



There is a good agreement between GRG orbit and external orbitsNo patch but small bias





There is a good agreement between GRG orbit and external orbits
No patch but small bias, opposite for JPL comparison

#### Comparison to CNES (POE-F) orbits Mapping radial orbit differences

Average (128 weeks) (from January 2017 to June 2019)

#### **REF = GRG orbit**



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There is a good agreement between GRG orbit and external orbits
No patch but small bias

# Comparison to CNES (POE-E/F), GSFC and JPL orbits Altimeter crossover Cycles

Altimeter Xover (in cm)			
Orbit	Jason-2	Jason-3	Sentinel-3A
GRG	5.70	5.58	5.48
CNES	5.66	5.56	5.44
GSC	X	5.57	X
JPL	X	5.55	X

The altimeter xover obtained with GRG orbit are at a good level

• The level is comparable but slightly higher to the others orbits evaluated



#### Impact on the new multi-satellite solution

□ Multi-satellite Solution (weekly) compared to DPOD2014 (from Jan. 2017 to Dec. 2018) GRG OLD serie /NEW serie





#### Summary

#### Preparation for the next ITRF

Models and Standards recommended by IERS and Processing strategy update

#### □ Status of POD for Jason-2/3 and Sentinel-3A/3B satellites

- The POD results are of good quality but the DORIS RMS for Jason-3 and Sentinel-3 satellites are still higher than the other DORIS satellites. It can be due from the RINEX format. For Jason-3, it can also be explained by the SAA effect.
- For the GRG DORIS-only orbit, the SLR residuals are still higher than those of the external orbits (CNES, GSFC and JPL)
- There is good agreement between GRG orbit and external orbits but there is still room for improvement

#### **Given Setup Setup**

- Finalize the Preparation to the next ITRF:
  - Implement the CNES downweighting data
  - Analyze Geocenter and Scale factor from single satellite solutions
  - Use quaternions for Sentinels and Cryosat-2 satellites
  - Mitigate the ~118-day draconitic signal identified in the DORIS geodetic products
  - Apply a strategy recommended by IDS to mitigate the SAA effect on the multi-satellite solution
- Comparison to GRG GPS-only orbit and external DORIS-only orbits



#### **Backup: South Atlantic Anomaly**

The South Atlantic Anomaly (SAA) is an area where the Earth's inner Van Allen radiation belt comes closest to the Earth's surface, dipping down to an altitude of 200 kilometres. This leads to an increased flux of energetic particles in this region and exposes orbiting satellites to higher-than-usual levels of radiation. The SAA is the near-Earth region where the Earth's magnetic field is weakest relative to an idealized Earth-centered dipole field. The increased radiation perturbs the crystal quartz oscillators that are the heart of the DORIS system, causing short-term and long-term changes in the frequency behavior





